

磨耗尖鼻負稜碳化鎢車刀切削碳纖複合材料(CFRP)

之溫度預測研究

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摘要

本研究是一種選用 K 型材質的尖鼻負稜主刃碳化鎢車刀片, 當刀尖在工具磨床上預 先磨有一磨耗量後,針對碳纖維複合材料(CFRP)做車削研究。研究當中除了量測三軸切 削力外,並利用切削過程中,刀面與工件之摩擦面積,以計算出摩擦力。其次配合有限 元素分析技術(FEA),利用 AbqusTM 軟體及逆向分析法 (Inverse method),以預測磨耗尖 鼻負稜碳化鎢車刀切削碳纖複材時其刀尖之表面溫度。最後與紅外線儀器所量測的結果 作比較,結果顯示量測值與預測值很接近。

關鍵詞:車削、切削溫度、碳纖複材、有限元素分析(FEA)

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Prediction of Cutting Temperatures in Turning Carbon-Fiber-Reinforce-Plastics Materials Using Chamfered Main Cutting Edge Sharp Tools Considering Wear

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ABSTRACT

Temperatures of the carbide tip's surface when turning Carbon-Fiber-Reinforced-Plastics (CFRP) materials with a K type sharp main cutting edge tool considered wear is investigated. The frictional forces and heat generated in the basic cutting tools are calculated by using the measured cutting forces and the theoretical cutting analysis. The heat partition factor between the tip and chip is solved by using the inverse heat transfer analysis, which utilizes temperature on the carbide tip's surface measured by infrared as the input. The carbide tip's surface temperature is determined by finite element analysis (FEA, AbaqusTM software) and compared with temperatures obtained from experimental measurements by infrared . Good agreement demonstrates the proposed model.

Keywords: Turning, cutting temperatures, CFRP, FEA.

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1. INTRODUCTION

Singamneni (2005) demonstrated the mixed finite and boundary element method (FEM) finally enables the estimation of the cutting temperatures which is a simple, efficient method, and at the same time it is quite easy to be implemented. The objective of this paper is to set up an oblique cutting CFRP model to study three-dimensional cutting temperature for a sharp worn tool with a chamfered main cutting edge.

2. THEORETICAL ANALYSIS

Bhatnagar et al. (1995) showed that in machining of fiber reinforced plastic (FRP) composite laminates; it can be assumed that the shear plane in the matrix depends only on the fiber orientation and not on the tool geometry. Chang (2008) presented a model to predict the cutting temperatures in turning of glass-fiber-reinforced plastics with chamfered main cutting edge sharp worn tools that can accurately predict the cutting temperatures and the cutting forces. For the case of chamfered main cutting edge, temperatures and forces depend on nose radius *R*, first side rake angle α_{S1} , second side rake angle α_{S2} , as shown in Fig.1 and Table 1.



 $C_{\bullet} = \begin{bmatrix} N_{\bullet} \\ R_{\bullet} \\ R_$

Figure 1. Basic model of the chamfered main edge.

Figure 2. Specifications of tool face with wear tool when wear $(f > R, R \neq 0)$.

2.1 Shear Area in the Cutting Process with Chamfered Cutting Edge Sharp Tools Considering Wear

Fig. 2 reveal that the geometrical specification of tool wear on the tool face (triangle *CNM*) can be derived from the values of t_W and φ_A when already measured.

$$A = A_1 + A_2 + A_3 + A_s$$
 (as shown in Fig.1) (1)

$$A_{1} = 0.5a_{3}b_{3}\sin\theta_{3} = 0.5a_{3}b_{3} [1 - (a_{3}^{2} + b_{3}^{2} - c_{3}^{2})/2a_{3}b_{3}]^{1/2}\}(A_{1} = \Delta NBE)$$
(2)

$$A_{2} = \frac{1}{2}(a_{4} + b_{4}) \cdot h_{4}(A_{2} = \text{rectangle } MDFE') \quad (3), \quad A_{3} = A_{31} + A_{32} \quad (A_{3} = \Delta ME'E + \Delta MNE) \quad (4)$$

$$A_{31} = \frac{a_5 b_5}{2\cos\phi_e} \sin(\frac{0.5\pi + \alpha_b - \angle A31}{2})$$
(5), $\angle A31 = \cos^{-1}[\frac{c_5^2 + d_5^2 - e_5^2}{2c_5 d_5}]$ (6)

$$A_{32} = g_5 h_5 \frac{\sin(\angle A32)}{2\cos\phi_e} \qquad (7)_{\angle A32 = \cos^{-1}[\frac{h_5^2 + n_5^2 - m_5^2}{2h_5 n_5}] - \sin^{-1}[\frac{l_5}{s_5}\sin(\frac{\pi}{2} - \alpha_b)]} \tag{8}$$

$$A_s = (0.5W_e^2 \cos^2 \alpha_{s1} \tan C_s) / (\cos \alpha_b \sin \phi_e) (A_s \text{ is the area of secondary chip:} \Delta D' \overline{YJ})$$
(9)

$$Q = Q_1 + Q_2 + Q_3 \tag{10}$$

$$Q_1 = \frac{0.5(d/\cos C_s - W_e \cos^2 \alpha_{s1} \tan C_s)}{\cos \alpha_b} \cdot \frac{f \cos C_s - W_e \cos \alpha_{s1}}{\cos \alpha_{s2}} - (\overline{CN} \cdot \overline{NM} \sin \phi_B) / 0.5$$
(11)

$$Q_2 = \frac{W_e \cos \alpha_{s1} (d / \cos C_{s1} \tan C_s)}{\cos \alpha_b} - \overline{CN} \cdot W_e \cos \alpha_{s1} \quad (12), \quad Q_3 = (0.5W_e^2 \cos \alpha_{s1} \tan C_s) / \cos \alpha_b \quad (13)$$

$$\overline{CM} = t_W (\cos C_S + \sin C_S \tan \phi_A) \qquad (14), \qquad \overline{CN} = \frac{t_W (\cos C_S + \sin C_S \cdot \tan \phi_A)}{(\sin \phi_A \tan \phi_A + \cos \phi_B)} (15)$$

$$\overline{NM} = (\overline{CM}^2 + \overline{CN}^2 - 2\overline{CM} \cdot \overline{NM} \cos \phi_B)^{\frac{1}{2}}$$
(16)

$$\angle CMN = \cos^{-1}\left[\frac{(\overline{CM}^2 + \overline{CN}^2 - \overline{NM}^2)}{2\overline{CM} \cdot \overline{CN}}\right] \quad (17), \angle CNM = \cos^{-1}\left[\frac{(\overline{CN}^2 + \overline{NM}^2 - \overline{CM}^2)}{2\overline{CN} \cdot \overline{NM}}\right] \quad (18)$$

The contact length of the tool edge can be considered as in Fig. 2.

$$\overline{NM}\cos(\frac{\pi}{2} - \angle CMN) < (\frac{f\cos C_s}{\cos \alpha_e} - W_e \cos \alpha_{s1})$$



Figure 3. Tool tip wears with chamfered Figure 4. Flow chart of the inverse heat transfer main cutting edge tool.

From the above diagram, the contact length is $l_f = \overline{HN} + \overline{NM} + \overline{MD} = \overline{i} \cdot \overline{ii} + \overline{ii} \cdot \overline{iii} + \overline{iii} \cdot \overline{iv} =$

$$\left[\frac{f\cos C_s / \cos \alpha_e - W_e \cos \alpha_{s1}}{\cos(C_e - C_s)} + \overline{NM} + \frac{d}{\cos C_s \cos \alpha_b} - \overline{CM}\right]$$
(19)
$$l_p = \overline{HN} \cos C_e + \overline{NM} \cos(\angle CNM - C_e) + \overline{MD} \sin C_s \left[\frac{f\cos C_s - W_e \cos \alpha_{s1}}{\cos \alpha_e \cos(C_e - C_s)} + \overline{NM} \cos(\angle CNM - C_e) + (d / \cos C_s - \overline{CM}) \sin C_s\right]$$
(20)

2.2. Energy Method to Predict Cutting Force

Transformation equations used to obtain the normal (N_s) and shear forces (F_s) along the fiber direction in terms of the principal (F_c) and thrust components (F_t) are shown in Eqs. (21) and (22).

$$N_{S} = F_{c} \sin \theta + F_{t} \cos \theta \qquad (21), \qquad F_{S} = F_{c} \cos \theta + F_{t} \sin \theta, \text{Liu} (2002) \qquad (22)$$

$$\tau_{S} = \tau_{compsite} = \tau_{fiber} V_{f} \text{ by Rosen and Dow(1987)} \quad (V_{f} \text{ is fiber contains})$$

$$V_{S} = V \cos \alpha_{e} / \cos(\varphi_{e} - \alpha_{e}), \qquad (23), \qquad f_{t} = \tau_{S} t_{1} \sin \beta / \cos(\varphi + \beta - \alpha) \sin \varphi \qquad (24)$$

$$V_{C} = V \sin \varphi_{e} / \cos(\varphi_{e} - \alpha_{e}) \qquad (25) \qquad \alpha_{e} = \sin^{-1} (\sin \alpha_{S2} \cos \alpha_{b} \cos \eta_{C} + \sin \eta_{C} \sin \alpha_{b}) (26)$$

Therefore, $(F_H)_{Umin}$ was determined by solving Equ. (28) in conjunction with the energy method by Reklaitis etc.(1984). (27)

$$F_{H=}\frac{U_{\min}}{V} = \left\{\frac{\tau_S \cos\alpha_e A}{\cos(\varphi_e - \alpha_e)} + \frac{\tau_S \sin\beta \cos\alpha_e Q}{\cos(\varphi_e + \beta - \alpha_e)\cos(\varphi_e - \alpha_e)}\right\}$$
(28), where the frictional force is

determined by
$$F_t = \frac{\tau_s \sin \beta \cos \alpha_e Q}{[\cos(\phi_e + \beta - \alpha_e) \sin \phi_e]}$$
 (29)

$$N_t = \frac{\left[(F_H) - (F_t)_{U_{\min}} \sin \alpha_e\right]}{\cos \alpha_{s2} \cos \alpha_b}$$
(30)

2.3. Calculation of Flank Wear

Thus, the flank wear V_B is a function of t_W , θ_e and α_e . $V_B = t_W \cos \alpha_e (\cot \theta_e - \tan \alpha_e)$ (31)

2.4. Finite Element Model

The finite element analysis software AbaqusTM is used in this study. The finite element mesh of the carbide tip is shown in Fig. 5, which was modeled by 58,000 four-node hexahedral elements. 8*6 nodes are located on the projected contact length between the tool and the workpiece, 3 * 6 nodes are located on the chamfered width of the main cutting edge, and 1*6 nodes are placed on flank wear.



Figure 5. Solid model of the chamfered edge tool.

Figure 6. Experimental set-up.

2.5. Modified Carbide Tip Temperature Model

Magnitude of the tip's load is shown in the following Eqs. (32) and (33)

$$K = U_f / A' \quad (32) \quad A' = L_p (d + W_e + V_b) \qquad (33), \quad \rho c \frac{\partial T}{\partial t} = k \frac{\partial^2 T}{\partial x^2} + k \frac{\partial^2 T}{\partial y^2} + k \frac{\partial^2 T}{\partial z^2} \qquad (34)$$

where ρ is the density, c is the thermal conductivity, and k is the heat capacity.

$$q_{tool} = Kq_f$$
 Li and Shih (2005) (35), $Obj(K) = \sum_{i=1}^{n_i} \sum_{j=1}^{n_j} (T_j^{t_i}|_{exp} - T_j^{t_i}|_{est})^2$ (2005) (36)

In this study, *K* is assumed to be a constant for all cutting edges. The inverse heat transfer method is used to find the value of *K* under certain turning speeds.

2.6. Inverse Heat Transfer Solution and Validation

The flowchart for inverse heat transfer solution of *K* was obtained by the AbaqusTM solver and is summarized in Fig. 5. The inverse heat transfer method is applied to solve *K* by minimizing an energy function on the tip surface determined by Eqs. (35)- (36) and finite element modeled temperature at specific infrared locations, as shown in Fig. 5 on the tip face. The discrepancy between the experimentally measured temperature by infrared pyrometer, j by time t_i , $T_j^{t_i}|_{exp}$ and finite element estimated temperature at the same infrared location and time, $T_j^{t_i}|_{ext}$ determines the value of the objective function.

3. EXPERIMENTAL METHODS AND PROCEDURES

Experimental set up is shown in Fig 6. The work material used was 0°; unidirectional filament wound fiber of CFRP with Vinylester resin composite materials in the form of bars having a diameter of 40 mm and 500 mm length by Liu (2002). Table 2 shows some of the physical and mechanical properties of CFRP prior to carrying out the cutting experiments. The cutting tools used in the experiments are Sandvik H1P (*K type*) by Brookes (1992). Tool composition: WC 85.5%, TiC 7.5%, Ta (Nb)C 1% and Co 6 %(30), HV = 1850, density=12.9 g/cm³, thermal conductivity = $60W/m^{-6}K$ and heat capacity= $235 J/Kg^{-6}K$.

4. RESULTS AND DISCUSSION

4.1 From Fig. 7, it proved that the cutting edge temperature of the chamfered main edge tool was lower than unchamfered main cutting edge tool.

4.2 According to Fig. 7, the tip temperatures of chamfered main cutting edge sharp worn tools were not high and the inverse (calculated) data correlates closely with the experimental values.

4.3 From Fig.7, the cutting temperatures of chamfered main cutting edge sharp worn tool is the lowest, when $C_s = 20^\circ$, $\alpha_{s1}(\alpha_{s2}) = -10^\circ(10^\circ)$ and the temperature is not exceed 350°C.

4.4 From Fig.7, it proved that the distribution of chamfered main cutting edge sharp worn tool's temperature was close the Fig. 8.

5. CONCLUSIONS

The test investigated the cutting forces and cutting temperature during the turning of CFRP. Chamfered main cutting edge sharp worn tools with C_s equals to 20 $\alpha_{s1}(\alpha_{s2}) = -10^{\circ}(10^{\circ})$ and nose radius R=0.3 mm, produce the lower cutting forces and lower cutting temperature. Good correlations between predicted values and experimental results of forces and temperatures during machining with chamfered main cutting edge sharp worn tools in cutting CFRP.

APPENDIX

Coefficients of the tool have a sharp corner (R = 0) with tool wear

$$a_{3} = \{\left[\frac{f\cos C_{s}}{\cos \alpha_{s2}} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s})\right]^{2} [\tan \eta_{c} - \tan(C_{e} - C_{s})]^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}(\cos^{2} \alpha_{e})^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}(\cos^{2} \alpha_{e})^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}(\cos^{2} \alpha_{e})^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}(\cos^{2} \alpha_{e})^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}(\cos^{2} \alpha_{e})^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}(\cos^{2} \alpha_{e})^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}(\cos^{2} \alpha_{e})^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}(\cos^{2} \alpha_{e})^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}(\cos^{2} \alpha_{e})^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}(\cos^{2} \alpha_{e})^{2} + (f\cos C_{s}/\cos \alpha_{e} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s}))/\cos \eta_{c}]^{2}$$

$$b_{3} = \left[\frac{\frac{f\cos C_{s}}{\cos \alpha_{s2}} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s})}{\cos \eta_{c}}\right] \frac{\cos \alpha_{e}}{\sin \phi_{e}} (2), \quad c_{3} = \left[\frac{f\cos C_{s} / \cos \alpha_{s2} - W_{e}\cos \alpha_{s1}}{\cos(C_{e} - C_{s})} - \overline{CN}\right] (3)$$

$$a_4 = d/\cos C_s - [f\cos C_s/\cos \alpha_{s2} - W_e\cos \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(\eta_c - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(\eta_c - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(\eta_c - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(\eta_c - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(\eta_c - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(\eta_c - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(\eta_c - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(\eta_c - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(\eta_c - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{NM}\sin \alpha_{s1} - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(Q_e - \tan(C_e - C_s)) - \overline{CN}\cos(C_e - C_s)]^2 [\tan \eta_c - \tan(C_e - C_s)]^2$$

$$(\pi - \angle CMN - \theta_B - \eta_c - C_e - C_s) / \sin(\theta_B - \eta_c + C_e - C_s) (4), b_4 = d / (\cos C_s - \cos C_e) - \overline{CM} (5); \quad h_4 = (c_4^2 - d_4^2)^{1/2} (6)$$

$$c_{4} = \left[\frac{f\cos C_{s} - W_{e}\cos\alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s})}{\cos\eta_{c}} + \left[\frac{\overline{NM}\sin(\pi - \theta_{B} - \angle CNM)}{\sin(\theta_{B} - \eta_{c} + C_{s} - C_{e})}\right]\frac{\cos\alpha_{e}}{\sin\varphi_{e}}$$
(7)

$$a_{5} = \overline{NM} \quad b_{5} = b_{3} \quad c_{5} = a_{3}, \quad e_{5} = \left[\frac{f \cos C_{s}}{\cos \alpha_{s1}} - W_{e} \cos \alpha_{s1} - \overline{CN} \cos(C_{e} - C_{s})\right] [\tan \eta_{c} - \tan(C_{e} - C_{s})]$$
(8)

$$d_{5} = \left[\frac{\left(\frac{f\cos C_{s}}{\cos \alpha_{e}} - W_{e}\cos \alpha_{s1} - \overline{CN}\cos(C_{e} - C_{s})\right)}{\cos \alpha_{e}}\left(\tan \alpha_{e} + \cot \phi_{e}\right)$$
(9)

$$g_{5} = \left[\frac{f\cos C_{s}/\cos \alpha_{s2} - W_{e}\cos \alpha_{s1} - CN\cos(C_{e} - C_{s})}{\cos \eta_{c}} + \left[\frac{\overline{NM}\sin(\pi - \theta_{B} - \angle CNM)}{\sin(\theta_{B} - \eta_{c} + C_{s} - C_{e})}\right]\frac{\cos \alpha_{e}}{\sin \varphi_{e}} \quad (10)$$

$$h_5 = (m_5^2 + n_5^2 - 2m_5 n_5 \sin \alpha_b)^{\frac{1}{2}}$$
(11)

$$m_{5} = \frac{NM\sin(\pi - \theta_{B} - \angle CNM + \eta_{c} - C_{e} + C_{s})}{\sin(\theta_{B} - \eta_{c} + C_{e} - C_{s})} \quad (12), \quad n_{5} = g_{5}\sin\phi_{e}(\tan\alpha_{e} + \cot\phi_{e}) - d_{5} \quad (13)$$

$$l_{5} = \left[\frac{f\cos C_{s}}{\cos \alpha_{S1}} - W_{e}\cos \alpha_{S1} - \overline{CN}\cos(C_{e} - C_{s})\right] (\tan \eta_{c} - \tan(C_{e} - C_{s}) - \overline{CM} + \frac{\overline{NM}\sin(\pi - \theta_{B} - \angle CNM + \eta_{c} - C_{e} + C_{s})}{\sin(\theta_{B} - \eta_{c} + C_{e} - C_{s})}$$
(14)

$$s_5 = (l_5^2 + r_5^2 - 2l_5r_5\sin\alpha_b)^{\frac{1}{2}} (15), \ r_5 = g_5\sin\varphi_e(\tan\alpha_e + \cot\varphi_e)/\cos\alpha_e$$
(16)

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side cutting	tool	side rake angles	Nose roundness (R)	carbide
edge angle C_s	no	$\alpha_{S1}, \alpha_{S2}(\alpha_{r1}, \alpha_{r2})$		tool
20 °	1	10°, -10° (10 ° , -10 °)	0.0, 0.1 (sharp and worn)	K10
20 °	2	30°, -30 ° (30 ° , -30 °)	0.0, 0.1 (sharp and worn)	K10
30 °	3	10°, -10 ° (10 ° , -10 °)	0.0, 0.1 (sharp and worn)	K10
30 °	4	30°, -30 ° (30 ° , -30 °)	0.0, 0.1 (sharp and worn)	K10
40 °	5	10°, -10 ° (10 ° , -10 °)	0.0, 0.1 (sharp and worn)	K10
40 °	6	30°, -30 ° (30° , -30 °)	0.0, 0.1 (sharp and worn)	K10
notation: tool holder & tips		Tool α_{s_1}		

Table 1. Tool geometry specifications (chamfered main cutting edge)

density g/cm ³	thermal conductivity kCal/hr [°] C	fiber contain	thermal expansion (10 ⁻⁶ / [°] C)	tensile strength (kg/cm ²)	compressive strength (kg/cm ²)	shear strength (kg/cm ²)	modulus tensile (kg/cm ²)
1.7~1.9	0,21~0.28	75%	2~9	3.5~4	3.5~3.9	1.5~2	235~400



Figure 7. Shows the cutting temperatures vs C_s for different values α_{s_1} and α_{s_2} with chamfered and unchamfered sharp tool at d=3.0 mm, f=.33 mm/rev, V=252m/min respectively.



Figure 8. Temperature distribution with chamfered cutting edge inserts (a) heat flux (b) near the tool nose at $C_s = 30^\circ$, $\alpha_{s1}(\alpha_{s2}) = -10^\circ(10^\circ)$, d=3.0 mm, f=0.33 mm/rev, and V=252 m/min (GFRP)